



THE AEROSPACE
CORPORATION

GUVI
Global Ultraviolet Imager
Critical Design Review



GUVI Thermal Design

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Agenda

- Action Item Status
- Changes Since PDR
- Temperature Requirements
- Design Overview
- Results
- Conclusions



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Thermal Action Item Status

- PDR AI 08 Closed
- PDR AI 09 Closed
- PDR AI 12 Closed
- PDR AI 13 Closed



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Changes Since PDR

- Additional radiator area added to SIS housing:
 - increased from 25 to 52 sqin to improve detector hot operation
- Scan motor thermally strapped to +Z deck
 - accomplished by using flexible copper straps and a 4x4 aluminum interface plate
- Orbit average heater power has increased
 - sunsafe (survival condition) has now been analyzed
 - operational has increased by 2 watts



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Operational Temperature Requirements

Component	Thermal Design Range	Flight Range	Test Range
Scan Motor Interface	-24 to +55	N/A	-24 to +55
Detectors	-10 to +15	-20 to +25	-25 to +30
SIS Housing*	-10 to +20	-20 to +30	-29 to +50
Electronic Box Interface	-24 to +55	N/A	-24 to +55

*will be tested at SSG
before detectors are
installed



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Survival Temperature Requirements

Component	Thermal Design Range	Flight Range	Test Range
Scan Motor Interface	-29 to +60	N/A	-29 to +60
Detectors	-20 to +40	-25 to +45	-30 to +50
SIS Housing*	-19 to +40	-24 to +45	-29 to +50
Electronic Box Interface	-29 to +60	N/A	-29 to +60

*will be tested at SSG before detectors are installed



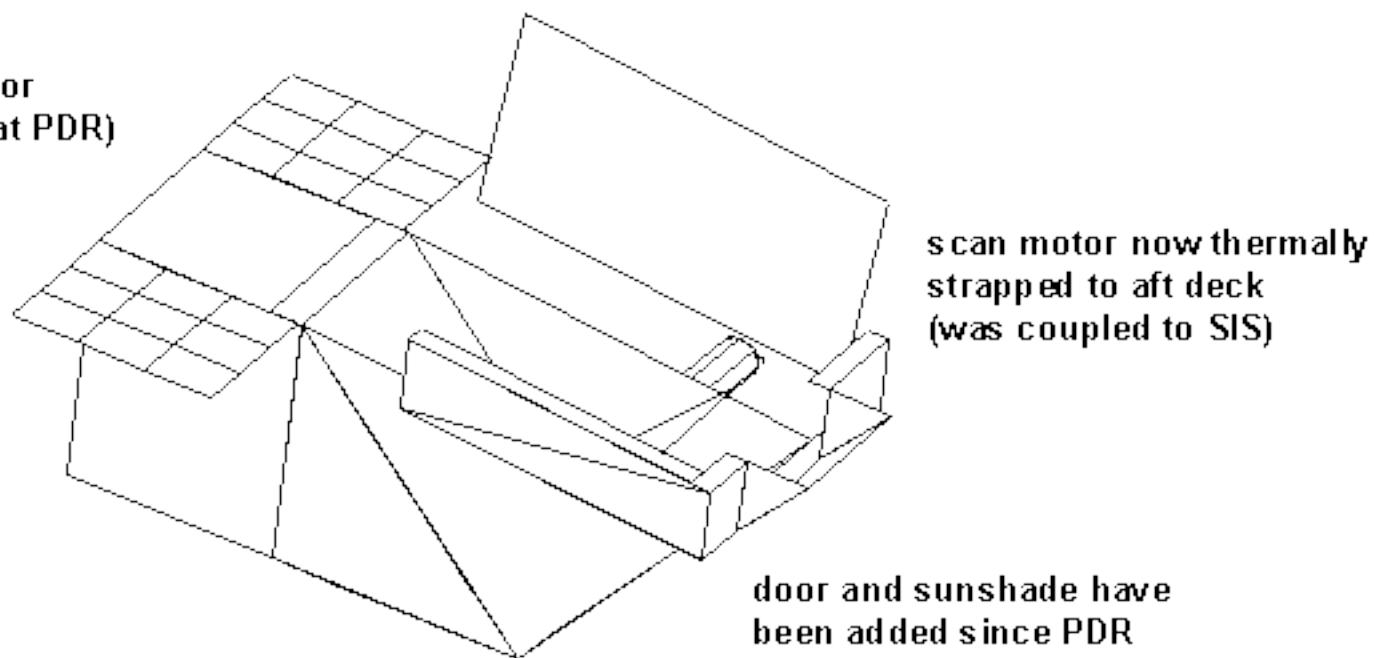
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Thermal Geometry Model

radiator now
52 sqin radiator
(was 22 sqin at PDR)





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Thermal Design Overview

- Survival Temperatures maintained by survival heater circuit during non-operation mode
- Heaters and thermostats maintain temperatures during cold operating conditions
- Flexible copper thermal strap couples scan motor case to +Z deck (8-9 °C/W assumed thermal resistance)
- Spectrograph is hard mounted to +Z deck but is thermally isolated due to titanium feet
- Black kapton MLI to cover designated areas on spectrograph
 - atomic oxygen effects will be mitigated by overcoating external kapton with 1300 angstroms of SiO₂ (GUVI-PDR-AI13)
 - Sheldahl standard process



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Analysis Parameters

Parameter	Hot Case	Cold Case
Solar Constant	450.0*	408.0*
Albedo Constant	0.4	0.2
Earth Constant	85.0*	60.0*
+Z Deck Temperature	+55 C	-29 C
Blanket Thru-Emittance	0.01	0.04

Notes: * Units are BTU/Hr-SqFt



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Analysis Parameters (2)

Component	Hot Operation Power (W)	Cold Operation Power (W)	Survival Power (W)
Scan Motor	4.000 *	2.000 *	0.0
Detector #1	0.100	0.000	0.0
Detector #2	0.100	0.000	0.0
Total:	4.20	2.000	0.0

Notes: * Scan Motor Hot Case assumes 100% Duty Cycle with peak power of 4.0 W ; Cold Case assumes a 50% Duty Cycle.



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Results Summary

- Maximum predicted SIS housing gradient:
 - 14 °C transient
 - 10 °C orbit average
- All operational and survival temperature predictions are within upper and lower limits
- Maximum orbit average heater power predictions
 - survival:
 - 11 watts sunsafe
 - 4 watts nominal spacecraft attitude
 - 7 watts operational

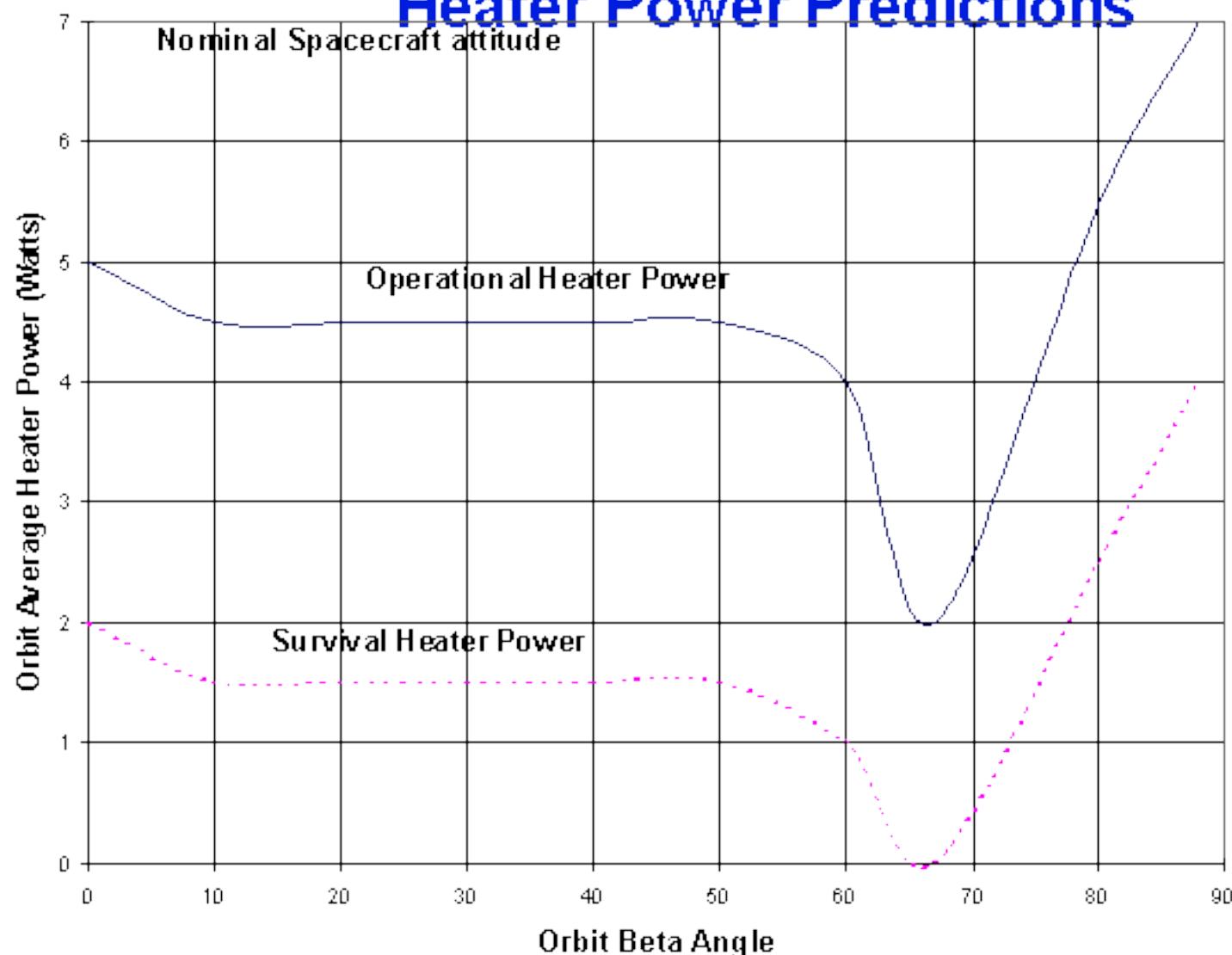


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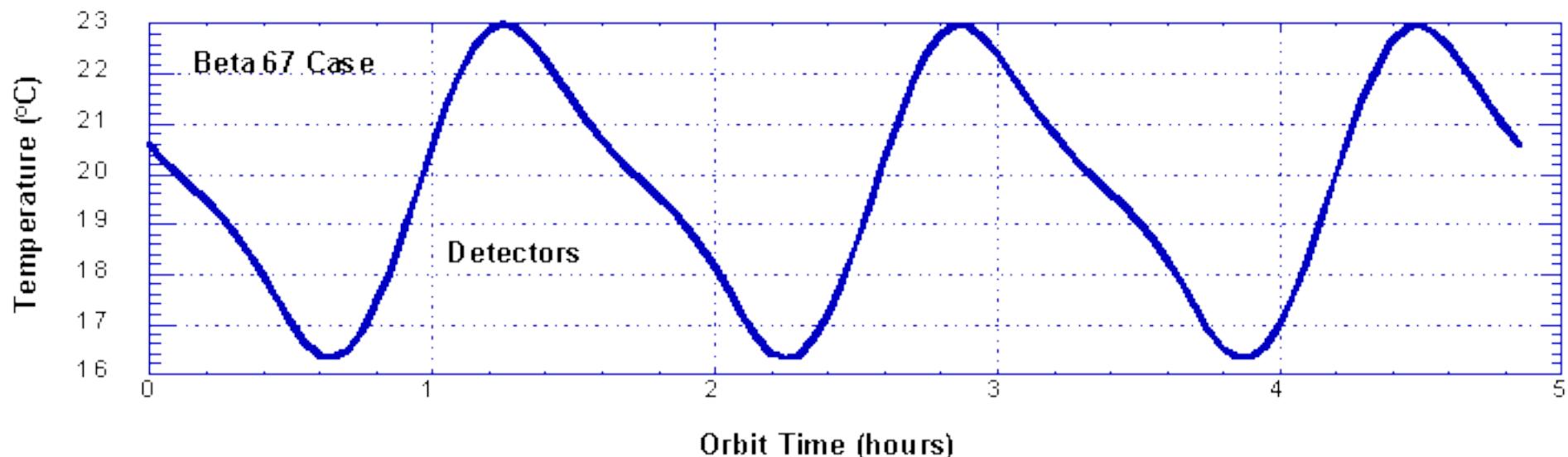
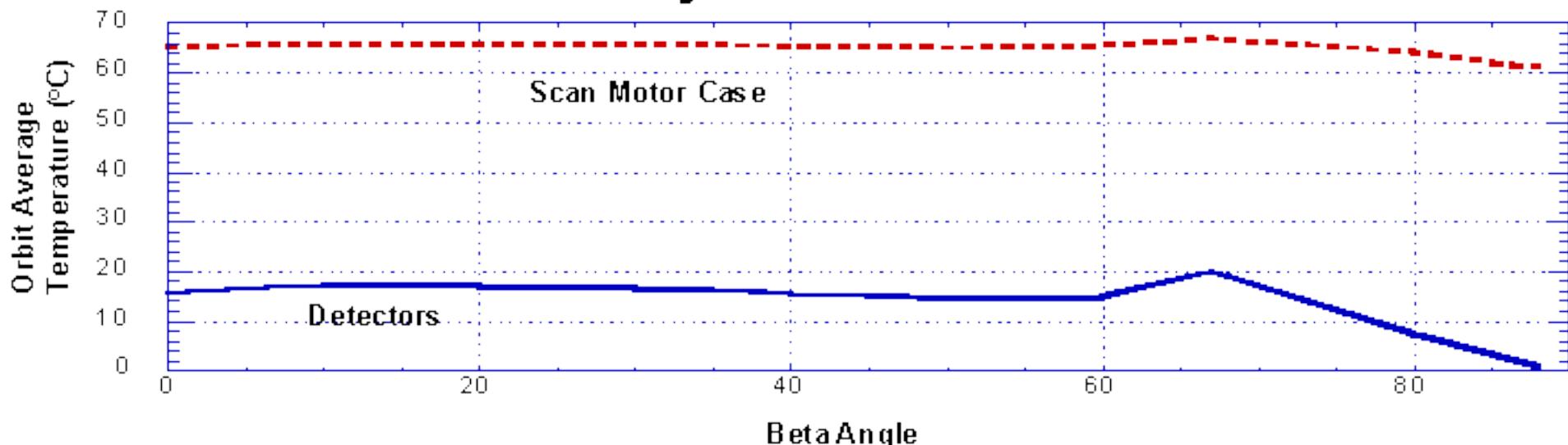


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Hot Analysis Results



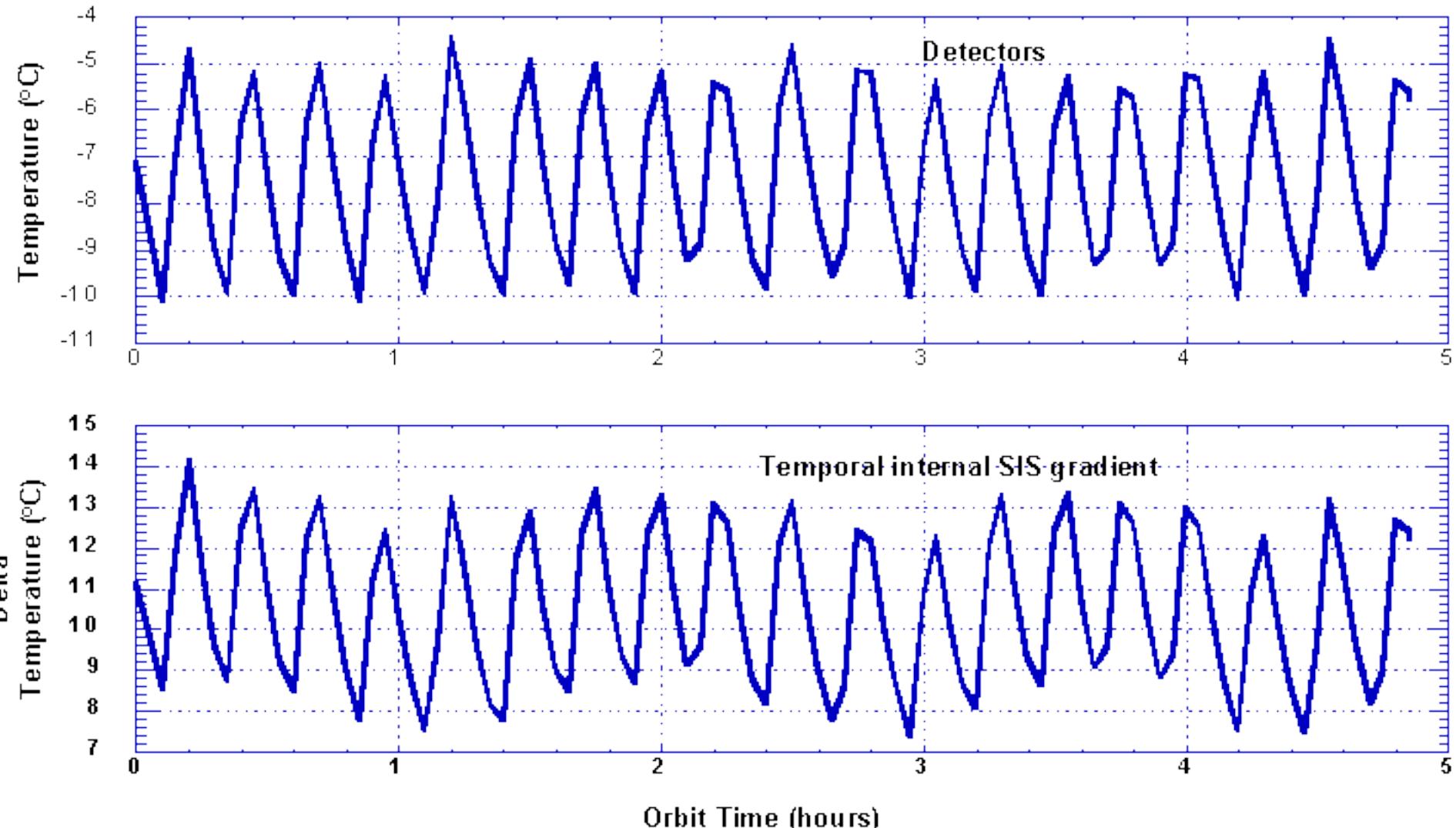


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Cold Analysis Results at Beta 88 Orbit Conditions





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Heater Circuit Summary

- Survival heater and thermostat:
 - single heater on SIS housing under detector #1 controlled by a thermostat near detector #2
 - maximum 11 watts orbit average; 26.6 watt peak at 35 V
 - 46.1 ohms
 - nominal thermostat range: close @ -28 °C / open @ -24 °C
 - heater and thermostat to be supplied and integrated by SSG
- Operational heaters and thermostats:
 - separate heater/thermostat on each tube
 - maximum 7 watts orbit average; 17 watts peak power at 35 V
 - parallel circuit resistance is 72 ohm; 144 ohm each heater
 - going to use NEAR spares for the thermostats (PI10709-6)
 - SN 00003: close @ -10.5 °C / open @ -5.8 °C
 - SN 00004: close @ -10.7 °C / open @ -5.9 °C



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Conclusions

- Testing
 - fully integrated thermal design and workmanship testing to start in October - November 1998 time frame
- Material Purchases
 - using NEAR spares for operational thermostats
 - purchase operational heaters and black kapton coated with SiO₂
- Reduced Thermal Models
 - completed and will be delivered after CDR